Pultrusion Conference 2021

Validation Testing of a Custom Pultruded FRP Shape for Long-span, Permanent Load Conditions

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ENGINEERING & CONSULTING







Meet your speaker

- Richard (Rich) Estes, PE
- BSCE University of Notre Dame [2013]
- Oilfield Engineer [2013-15]



- Public Works Engineer [2015-17]
- Structural Engineer [2017-current] <- Designing with FRP since 2017



Objectives

- Applications and needs for custom pultruded FRP shapes
- Case study of a validation testing program
- Statistical analysis worked example
- Lessons learned



Applications and Needs



Mission Critical









Industrial







Equipment Access and Support









FRP Industry Needs

- Seismic design
- Moment resisting connections
- Diaphragm action
- Longer spans





Case Study



Mission Critical – Long Span Permanent Load

- Steel primary frame
- FRP infill beams
- Span \geq 20 ft
- Dead load = 30-160 psf (equipment)
- Beam spacing \geq 4 ft
- Deflection limit = L/240





Optimization Limits

- Optimize cross-sectional area
- Limit depth and width to 18"
- Non-uniform thickness permissible
- Maximum thickness of 1"





 Strength and deflection checked per the provisions of the LRFD Pre-Standard



Preliminary Results

| Uniform Load [psf] | Spacing [ft] | Span [ft] | Unbraced Span [ft] | Optimized Beam Size [depthxflange widthxthickness] | Beam Area [in^2] | Optimized Steel Beam [W#x#] | Uniforr Load [p: | Spacing | Span [ft] | Unbraced Span [ft] | Optimized Beam Size [depthxflange widthxthickness] | Beam Area [in^2] | Optimized Steel Beam [W#x#] | ı | Uniform Load (psf) | Spacing [ft] | Span [ft] | Unbraced Span [ft] | Optimized Beam Size [depthxflange widthxthickness] | Beam Area [in^2] | Optimized Steel Beam [W#x#] |
|-----------------------|--------------|--------------|-----------------------|--|------------------------|-----------------------------------|---------------------|---------|--------------|-----------------------|--|------------------------|-----------------------------------|---|-----------------------|-----------------|-----------|-----------------------|--|------------------------|-----------------------------------|
| 160 | 6 | 20 | 20 | 18x15x7/8 | 40.5 | 12x30 | 14 | 0 6 | 25 | 25 | n/a | | 14x43 | | 160 | 6 | 20 | 10 | 18x10x3/4 | 27.4 | 14x22 |
| 160 | 5.5 | 20 | 20 | 16x15x7/8 | 38.8 | 8x28 | 16 | 0 5.5 | 25 | 25 | n/a | | 10x39 | | 160 | 5.5 | 20 | 10 | 18x10x3/4 | 27.4 | 14x22 |
| 160 | 5 | 20 | 20 | 18x15x13/16 | 37.7 | 8x28 | 10 | 0 5 | 25 | 25 | 18×18×1 | 52 | 10x39 | | 160 | 5 | 20 | 10 | 16x10x3/4 | 25.9 | 12×22 |
| 160 | 4.5 | 20 | 20 | 16x15x13/16 | 36.1 | 12×26 | 10 | 0 4.5 | 25 | 25 | 18×18×1 | 52 | 10×39 | | 160 | 4.5 | 20 | 10 | 14x10x3/4 | 24.4 | 8x21 |
| 160 | 4 | 20 | 20 | 18x14x3/4 | 33.4 | 8x24 | 10 | 0 4 | 25 | 25 | 18x18x15/16 | 48.9 | 10×33 | | 160 | 4 | 20 | 10 | 16x10x11/16 | 23.8 | 8x18 |
| | | | | | | | | | | | | | | | | | | | | | |
| 100 | 6 | 20 | 20 | 16x14x3/4 | 31.9 | 8x24 | 10 | 0 6 | 25 | 25 | 18x17x15/16 | 47 | 10x33 | | 100 | 6 | 20 | 10 | 12x10x3/4 | 22.9 | 8x18 |
| 100 | 5.5 | 20 | 20 | 16x14x3/4 | 31.9 | 8x24 | 10 | 0 5.5 | 25 | 25 | 18x17x7/8 | 44 | 8x31 | | 100 | 5.5 | 20 | 10 | 18x9x5/8 | 21.7 | 8x18 |
| 100 | 5 | 20 | 20 | 14x14x3/4 | 30.4 | 8x24 | 10 | 0 5 | 25 | 25 | 16x17x7/8 | 42.2 | 8x31 | | 100 | 5 | 20 | 10 | 16x9x5/8 | 20.5 | 10x17 |
| 100 | 4.5 | 20 | 20 | 12x14x3/4 | 28.9 | 10x22 | 10 | 0 4.5 | 25 | 25 | 18x17x13/16 | 40.9 | 8x31 | | 100 | 4.5 | 20 | 10 | 14x9x5/8 | 19.2 | 10x17 |
| 100 | 4 | 20 | 20 | 18×13×5/8 | 26.8 | 6x20 | 10 | 0 4 | 25 | 25 | 16x16x13/16 | 37.7 | 8×28 | | 100 | 4 | 20 | 10 | 12x9x5/8 | 18 | 8x15 |
| | | | | | | | | | | | | | | | | | | | | | |
| 60 | 6 | 20 | 20 | 16x13x5/8 | 25.5 | 6x20 | | 0 6 | 25 | 25 | 18x16x3/4 | 36.4 | 8x28 | | 60 | 6 | 20 | 10 | 12x9x5/8 | 18 | 10x15 |
| 60 | 5.5 | 20 | 20 | 14x13x5/8 | 24.2 | 6x20 | | 0 5.5 | 25 | 25 | 16x16x3/4 | 34.9 | 8x24 | | 60 | 5.5 | 20 | 10 | 14x9x9/16 | 17.4 | 12x14 |
| 60 | 5 | 20 | 20 | 18x12x9/16 | 23 | 8x18 | | 0 5 | 25 | 25 | 18x15x11/16 | 32.1 | 8x24 | | 60 | 5 | 20 | 10 | 18x8x1/2 | 16.5 | 8x13 |
| 60 | 4.5 | 20 | 20 | 16x12x9/16 | 21.9 | 8×18 | | 0 4.5 | 25 | 25 | 16x15x11/16 | 30.7 | 8x24 | | 60 | 4.5 | 20 | 10 | 16x8x1/2 | 15.5 | 8x13 |
| 60 | 4 | 20 | 20 | 18x12x1/2 | 20.5 | 6×15 | | 0 4 | 25 | 25 | 16x15x11/16 | 30.7 | 8x24 | | 60 | 4 | 20 | 10 | 18x8x7/16 | 14.5 | 10x12 |
| | | | | | | | | | | | | | | | | | | | | | |
| 50 | 6 | 20 | 20 | 18x12x9/16 | 23 | 8x18 | | 0 6 | 25 | 25 | 18x15x11/16 | 32.1 | 8x24 | | 50 | 6 | 20 | 10 | 18x8x1/2 | 16.5 | 8×13 |
| 50 | 5.5 | 20 | 20 | 16x12x9/16 | 21.9 | 8x18 | | 0 5.5 | 25 | 25 | 18x15x11/16 | 32.1 | 8x24 | Г | 50 | 5.5 | 20 | 10 | 16x8x1/2 | 15.5 | 8x13 |
| 50 | 5 | 20 | 20 | 16x12x9/16 | 21.9 | 6x15 | | 0 5 | 25 | 25 | 16x15x11/16 | 30.7 | 8x24 | Г | 50 | 5 | 20 | 10 | 14x8x1/2 | 14.5 | 10x12 |
| 50 | 4.5 | 20 | 20 | 18x12x1/2 | 20.5 | 6x15 | | 0 4.5 | 25 | 25 | 18x15x5/8 | 29.2 | 8x24 | | 50 | 4.5 | 20 | 10 | 18x8x7/16 | 14.5 | 10x12 |
| 50 | 4 | 20 | 20 | 16x12x1/2 | 19.5 | 6x15 | | 0 4 | 25 | 25 | 16x15x5/8 | 27.9 | 6x20 | | 50 | 4 | 20 | 10 | 16x8x7/16 | 13.5 | 8x10 |
| | | | | | | | | | | | | | | Г | | | | | | | |
| 40 | 6 | 20 | 20 | 18x12x1/2 | 20.5 | 6x15 | | 0 6 | 25 | 25 | 16x15x11/16 | 30.7 | 8x24 | | 40 | 6 | 20 | 10 | 18x8x7/16 | 14.5 | 10×12 |
| 40 | 5.5 | 20 | 20 | 18x12x1/2 | 20.5 | 6x15 | | 0 5.5 | 25 | 25 | 18x15x5/8 | 29.2 | 10x22 | | 40 | 5.5 | 20 | 10 | 12x8x1/2 | 13.5 | 10x12 |
| 40 | 5 | 20 | 20 | 16x12x1/2 | 19.5 | 6x15 | | 0 5 | 25 | 25 | 16x15x5/8 | 27.9 | 6x20 | Г | 40 | 5 | 20 | 10 | 16x8x7/16 | 13.5 | 8x10 |
| 40 | 4.5 | 20 | 20 | 14x12x1/2 | 18.5 | 6x15 | | 0 4.5 | 25 | 25 | 18x14x9/16 | 25.3 | 6x20 | | 40 | 4.5 | 20 | 10 | 14x8x7/16 | 12.74 | 8×10 |
| 40 | 4 | 20 | 20 | 18x11x7/16 | 17.2 | 6x15 | | 0 4 | 25 | 25 | 18x14x9/16 | 25.3 | 6x20 | | 40 | 4 | 20 | 10 | 18x7x3/8 | 11.72 | 6x9 |
| | | | | | | | | | | | | | | Г | | | | | | | |
| 30 | 6 | 20 | 20 | 14x12x1/2 | 18.5 | 6x15 | 1 | 0 6 | 25 | 25 | 18x14x9/16 | 25.3 | 6x20 | | 30 | 6 | 20 | 10 | 14x8x7/16 | 12.74 | 8×10 |
| 30 | 5.5 | 20 | 20 | 18x11x7/16 | 17.2 | 6x15 | | 0 5.5 | 25 | 25 | 18x14x9/16 | 25.3 | 6x20 | | 30 | 5.5 | 20 | 10 | 14x8x7/16 | 12.74 | 8×10 |
| 30 | 5 | 20 | 20 | 16x11x7/16 | 16.3 | 6x15 | | 0 5 | 25 | 25 | 16x14x9/16 | 24.2 | 6x20 | | 30 | 5 | 20 | 10 | 18x7x3/8 | 11.72 | 6x9 |
| 30 | 4.5 | 20 | 20 | 16x11x7/16 | 16.3 | 10x15 | 1 | 0 4.5 | 25 | 25 | 18x13x1/2 | 21.5 | 8x18 | | 30 | 4.5 | 20 | 10 | 16x7x3/8 | 10.97 | 6x8.5 |
| 30 | 4 | 20 | 20 | 14x11x7/16 | 15.4 | 6x12 | | 0 4 | 25 | 25 | 18x13x1/2 | 21.5 | 6x15 | | 30 | 4 | 20 | 10 | 14x7x3/8 | 10.22 | 6x8.5 |



Load Criteria Refinement

- Dead load = 60 psf
- Beam spacing = 5 ft
- Span = 24 ft
- Unbraced span = 8 ft
- Final shape = $18 \times \frac{1}{2} \times 8 \times \frac{3}{4}$





Why Test?

- Outperform the code prescribed capacity equations
- Establish performance confidence















Test Results

- Apparent stiffness, EI = 3.4×10^6 k-in²
- Deflection, Δ :
 - Before testing, predicted $\Delta = 0.87$ "
 - After testing, $\Delta = 0.7$ "



http://wiki.dtonline.org/index.php/Beam_Deflection



Statistics Worked Example



Referenced Standards

Pre-Standard for Load & Resistance Factor Design (LRFD) of Pultruded Fiber Reinforced Polymer (FRP) Structures (Final)

Submitted to: American Composites Manufacturers Association (ACMA)

November 9, 2010



This international standard was developed in accordance with internationally recognized principles on standardization established in the Decision on Principles for the Development of International Standards, Guides and Recommendations issued by the World Trade Oreanization Technical Barriers to Trade (TBT) Committee.



Designation: D7290 - 06 (Reapproved 2017)

Standard Practice for Evaluating Material Property Characteristic Values for Polymeric Composites for Civil Engineering Structural Applications¹

This standard is issued under the fixed designation D7290; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (e) indicates an editorial change since the last revision or reapproval.

1. Scope

1.1 This practice covers the procedures for computing characteristic values of material properties of polymeric composite materials intended for use in civil engineering structural applications. The characteristic value is a statistically-based material property representing the 80 % lower confidence bound on the 5th-percentile value of a specified population. Characteristic values determined using this standard practice can be used to calculate structural member resistance values in design codes for composite civil engineering structures and for establishing limits upon which qualification and acceptance criteric can be based.

1.2 This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.

1.3 This international standard was developed in accordance with internationally recognized principles on standardization estabilised in the Decision on Principles for the Development of International Standards, Guides and Recommendations issued by the World Irade Organization Technical Barriers to Trade (TBT) Committee.

2. Referenced Documents

2.1 ASTM Standards:² D883 Terminology Relating to Plastics

D3878 Terminology for Composite Materials D5055 Specification for Establishing and Monitoring Structural Capacities of Prefabricated Wood I-Joists

D5457 Specification for Computing Reference Resistance of

- ¹ This practice is under the jurisdiction of ASTM Committee D30 on Composite Materials and is the direct responsibility of Subcommittee D30.10 on Composites for Civil Structures.
- Current edition approved Aug. 1, 2017. Published September 2017. Originally approved in 2006. Last previous edition approved in 2011 as D7290-06(2011). DOI: 10.1520/D7290-06R17. ² For referenced ASTM standards, visit the ASTM website, www.astm.org, or

¹ For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For Annual Book of ASTM Standards volume information, refer to the standard's Document Summary page on the ASTM website.

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Structural Joints E6 Terminology Relating to Methods of Mechanical Testing E178 Practice for Dealing With Outlying Observations

E456 Terminology Relating to Quality and Statistics 2.2 Other Document:

MIL-Handbook-17 Polymer Matrix Composites, Volume 1, Revision F³

3. Terminology

3.1 Definitions—Terminology D3878 defines terms relating to high-modulus fibers and their composites. Terminology E66 defines terms relating to mechanical testing. Terminology E66 defines terms relating to statistics. In the event of a conflict between terms, Terminology D3878 shall have precedence over the other documents.

3.2 Definitions of Terms Specific to This Standard: 3.2.1 characteristic value—a statistically-based material property representing the 80% lower confidence bound on the 5th-percentile value of a specified population. The characteristic value accounts for statistical uncertainty due to a finite sample size.

3.2.1.1 Discussion—The 80 % confidence bound and 5thpercentile levels were selected so that composite material characteristic values will produce resistance factors for Load and Resistance Factor Design similar to those for other civil engineering materials (see Refs 1 and 2).⁴

3.2.1.2 Discussion—The term "characteristic value" is analogous to the term "basis value" used in the aerospace industry where A- and B-basis values are defined as the 95 %

³ Available from U.S. Government Printing Office Superintendent of Documents, 732 N. Capitol St., NW, Mail Stop: SDE, Washington, DC 20401, http://

www.uccess.gpo.gov. ⁴ The boldface numbers in parentheses refer to the list of references at the end of this standard.

Pre-Standard Section 2.4.3

2.4.3 Statistical Basis for Reference Strength and Stiffness

The reference strength and stiffness shall be determined in accordance with ASTM D7290. A minimum of 10 samples shall be tested to determine the reference strength or stiffness.

(a) Reference strength. The strength of pultruded FRP composite structural members and components shall be assumed to be described by a two-parameter Weibull distribution. The reference strength shall equal the characteristic value, defined at the 80% lower confidence interval on the 5th-percentile of the Weibull distribution.

(b) **Reference stiffness.** The elastic modulus in the longitudinal direction and in-plane shear modulus shall be described by two-parameter Weibull distributions.

(1) Strength and stability. The reference stiffness shall equal the characteristic value of the governing Weibull distribution.

(2) Structural analysis. The reference stiffness shall equal the mean value of the governing Weibull distribution.



ASCE

ASTM D7290

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- 1. Mean and standard deviation
- 2. Outlying observations
- 3. Material property value probability distribution
- 4. Maximum likelihood parameter estimates and coefficient of variation
- 5. Nominal value (two-parameter Weibull distribution)

6. <u>Characteristic value (80% confidence bound, 5th-percentile value)</u>



1. Calculate the Mean and Standard Deviation

- Sample mean, $\bar{x} = 4,051,870 \ k in^2$
- Sample standard deviation, $s_{n-1} = 252,524 k in^2$

2. Detection of Outlying Observations

- Outliers can skew the results
- Additional investigation may be required

3. Assumed Material Property Distribution

•
$$f(x) = \left(\frac{\beta}{\alpha}\right) \left(\frac{x}{\alpha}\right)^{\beta-1} exp\left[-\left(\frac{x}{\alpha}\right)^{\beta}\right]$$



4a. Maximum Likelihood Parameter Estimation

$$\cdot \frac{\sum_{i=1}^{n} x_i^{\widehat{\beta}} ln(x_i)}{\sum_{i=1}^{n} x_i^{\widehat{\beta}}} - \frac{1}{\widehat{\beta}} - \frac{1}{n} \sum_{i=1}^{n} ln(x_i) = 0$$

• Adjust $\hat{\beta}$ until equation is solved, $\hat{\beta}$ = 22.5

•
$$\hat{\alpha} = \left(\frac{\sum_{i=1}^{n} x_i^{\hat{\beta}}}{n}\right)^{\frac{1}{\hat{\beta}}} = 4,150,373$$



4b. Coefficient of Variation

•
$$COV = \frac{\sqrt{\Gamma\left(1+\frac{2}{\widehat{\beta}}\right) - \Gamma^2\left(1+\frac{1}{\widehat{\beta}}\right)}}{\Gamma\left(1+\frac{1}{\widehat{\beta}}\right)} = 0.055$$

5. Nominal Value

•
$$x_{0.05} = \hat{\alpha} * [0.0513]^{\frac{1}{\widehat{\beta}}} = 3,637,132 \ k - in^2$$





6. Characteristic Value

| n | 0.05 | 0.10 | 0.15 | 0.20 | 0.25 | 0.30 | 0.40 | 0.50 |
|------------|-------|-------|-------|-------|-------|-------|-------|-------|
| 10 | 0.950 | 0.899 | 0.849 | 0.800 | 0.752 | 0.706 | 0.619 | 0.541 |
| 11 | 0.953 | 0.906 | 0.860 | 0.814 | 0.769 | 0.725 | 0.642 | 0.567 |
| 12 | 0,956 | 0.913 | 0.869 | 0.826 | 0.783 | 0.741 | 0.662 | 0.589 |
| 13 | 0.959 | 0.918 | 0.876 | 0.835 | 0.795 | 0.755 | 0.679 | 0.609 |
| 14 | 0.961 | 0.922 | 0.883 | 0.844 | 0.805 | 0.767 | 0.694 | 0.626 |
| 15 | 0.963 | 0.926 | 0.889 | 0.851 | 0.814 | 0.778 | 0.707 | 0.641 |
| 16 | 0.965 | 0.929 | 0.894 | 0.858 | 0.822 | 0.787 | 0.719 | 0.655 |
| 18 | 0.968 | 0.935 | 0.902 | 0.869 | 0.836 | 0.803 | 0.739 | 0.678 |
| 20 | 0.970 | 0.940 | 0.909 | 0.878 | 0.847 | 0.816 | 0.755 | 0.698 |
| 22 | 0.972 | 0.944 | 0.914 | 0.885 | 0.856 | 0.827 | 0.769 | 0.714 |
| 24 | 0.974 | 0.947 | 0.919 | 0.891 | 0.864 | 0.836 | 0.781 | 0.728 |
| 26 | 0.975 | 0.949 | 0.923 | 0.897 | 0.870 | 0.844 | 0.791 | 0.741 |
| 28 | 0.976 | 0.952 | 0.927 | 0.902 | 0.876 | 0.851 | 0.800 | 0.752 |
| 30 | 0.977 | 0.954 | 0.930 | 0.906 | 0.882 | 0.857 | 0.809 | 0.761 |
| 32 | 0.978 | 0.956 | 0.933 | 0.910 | 0.886 | 0.863 | 0.816 | 0.770 |
| 34 | 0.979 | 0.957 | 0.935 | 0.913 | 0.890 | 0.868 | 0.822 | 0.778 |
| 36 | 0.980 | 0.959 | 0.938 | 0.916 | 0.894 | 0.872 | 0.828 | 0.785 |
| 38 | 0.980 | 0.960 | 0.940 | 0.919 | 0.897 | 0.876 | 0.833 | 0.791 |
| 40 | 0.981 | 0.962 | 0.942 | 0.921 | 0.901 | 0.880 | 0.838 | 0.797 |
| 42 | 0.982 | 0.963 | 0.943 | 0.924 | 0.904 | 0.883 | 0.843 | 0.803 |
| 44 | 0.982 | 0.964 | 0.945 | 0.926 | 0.906 | 0.886 | 0.847 | 0.808 |
| 46 | 0.983 | 0.965 | 0.946 | 0.928 | 0.909 | 0.889 | 0.851 | 0.813 |
| 48 | 0.983 | 0.966 | 0.948 | 0.929 | 0.911 | 0.892 | 0.854 | 0.817 |
| 50 or more | 0.984 | 0.967 | 0.949 | 0.931 | 0.913 | 0.895 | 0.858 | 0.821 |



Lessons Learned



Testing

- Need a significant testing apparatus in size and strength
- Adjustable support frame for varying beam lengths
- More tests increase confidence leading to a better performance





Installation

- Steel erection crew not required
- 4-5 person install crew
- Experienced install time of 10 minutes per beam
- Equivalent steel beam = W14x30 (no lateral bracing)
- Significant shipping cost reduction due to the lesser weight of the infill beams
- Savings of approximately 5% of the total structure weight
- Easy to maneuver within the existing space
- Ability to match drill bracing connections in the field



Questions?

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