

North American Pultrusion Conference

### Determination of real life aging on construction products made of glass fibre reinforced plastics (GFRP)

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- 2. Factors on Aging of GFRP
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- 5. Case Study 3: GFRP-Pipe at Power Plant
- 6. Approach for Failure Determination during Aging Test and Aging





# History: GFRP Antenna Carrier for a Radio

#### Monitoring of the GFRP structure by:



- Samples exposed at antenna ("witnesses") over a period of 24 years
- Final testing at samples cut out of GFRP antenna 30 years after installation

Television tower of Dresden, built 1969





Source: IMA Dresden, AVK Conference 1999

### **Factors on Aging of GFRP**

#### Long-term behaviour

- For the construction industry GFRP-components are designed for service lives of up to 100 years.
  - During service live: changing material properties must be taken into account
    - Suitable approach from standard DIN EN 13121-3: Factors of influence:
- A<sub>1</sub>: **Manufacturing** on the laminate (for advanced design: A<sub>1</sub> = 1.0)
- A<sub>2</sub>: Medium or environment on the laminate
- A<sub>3</sub>: Design **temperature** on the laminate
- A<sub>4</sub>: **Dynamic** (cyclic) **load** on the laminate
- A<sub>5</sub>: **Load duration** on the laminate
- A<sub>6</sub>: lonising **radiation**
- → Goal for design:  $E_d \le R_d$

$$E_d$$
 ... Design value of the stresses

- R<sub>d</sub>... Load-bearing resistance
- *E*<sub>i</sub> ... Load effect
- *R*<sub>k</sub> ... Laminate resistance
- $\gamma_{\rm F}$  ... Portion factor of variable impacts
- $\gamma_{M}$  ... Portion factor for material (= 1.4)



GFRP profiles in cooling tower









- 1<sup>st</sup> road bridge with GFRP structure in Germany
- Built in 2004 with asset profiles of manufacturer Fiberline Composites A/S
- Pre-assembly at manufacturer, final installation with on-site





GFRP-bridge: During installation; Source: Fiberline A/S





GFRP-bridge after installation; Source: Fiberline A/S



Adhesive bonding of GFRP-bridge; Source: Fiberline A/S



#### **IMA Services:**

- Materials and Component testing
- Manufacturer inspection with Inspection of pre-assembly
- Inspection of insertion and adhesion
- Measurement of the bridge deformation at load of 20 t
- "Witness" samples for long-term monitoring



Witness samples inside the bridge; Source: IMA







GFRP-bridge: After installation with Truck (weight approx. 20 t); Source: IMA

#### Test plan of witness samples:

Test / year	2006	2008	2009	2011	2015	2019	2023	2028	2033	2038
DLS	х	х	Х	х	Х	х	Х	Х	Х	х
Bending	х		Х	Х	Х	х	Х	Х	Х	х
Creep				Х		х		х		х

#### **Bonding properties for practice-aged samples:**

- Double lap shear test (DLS) following ASTM D 3528
- Test until shear failure of adhesive





DLS-samples before the test



DLS-samples after the test





#### Material properties for practice-aged samples:

- Samples taken from tested DLS-samples
- 3-point bending test following DIN EN ISO 14125
- Test until sample failure



ž

p





long-term monitoring of GFRP bridge: bending strength of laminate (DIN EN ISO 14125)







2006

0,0





### Stiffness properties for practice-aged samples:

- Creep test
- 3-point bending
- Test duration: 1000 h
- Load level: 35% of static strength
- Approximation using: 3-parameter-exponentiation approach acc. Findley
  - y variable for elongation (mm)
  - x variable for test time (h)
  - $k_1, k_2, k_3$  constants



Creep strain time diagram: Sampling 2011 Creep strain time diagram: Sampling 2019 0,1 0,1 0.01 0.1 100000 0.01 0.1 10 100 10000 10 100 1000 10000 100000 time [h] time [h] B058.06-2016-16 A B058.06-2016-17 B058.06-2016-18 B058.06-2009-B-7 B058.06-2009-B-8 B058.06-2009-B-9 - B058.06-2016-16 appro B058.06-2016-17 appro B058.06-2016-18 approx - B058.06-2009-B-7 appro - B058.06-2009-B-8 appro B058.06-2009-B-9 approx Creep modulus time diagram: Sampling 2011 Creep modulus time diagram: Sampling 2019 100.000 100.000 ------]S 10.000 10.000 1 000 1 000 0.0 100 10000 100000 0.01 0.1 100 1000 10000 100000 time [h] time [h] B058.06-2016-17 B058.06-2009-B-7 B058.06-2009-B-8 B058.06-2009-B-9 B058.06-2016-16 B058.06-2016-18 - B058.06-2009-B-7 approx - B058.06-2009-B-8 approx - B058.06-2009-B-9 approx B058.06-2016-16 approx. - B058.06-2016-17 approx. -- B058.06-2016-18 approx

- 1<sup>st</sup> cooling tower with GFRP structure in Germany (Wet cooling tower with forced venting system)
- Built in 1999 with regular profiles of manufacturer Fiberline Composites A/S
- After approx. 20 years of service life, square-tube GFRP profile 100 x 100 x 8 mm were removed from the structure for testing





GFRP profile 100 x 100 x 8 mm with drill holes and markings for sampling





Structure of cooler tower with GFRP-profiles; Source: Fiberline A/S



Cooling tower of pulp and paper mill; Source: Fiberline A/S



#### Material Testing at Standard Climate of:

Tensile test perpendicular to profile direction following DIN EN ISO 527-4





Compression test parallel to profile direction









Microscopy in the Area Surrounding the Drill Holes





Cutting depth approx. 1,4 mm





Cutting depth approx. 5,2 mm



- Crack depth is generally not large
- Negative influence of the cracks and delaminations on the load-bearing capacity of the profiles can be excluded
- Growth of the cracks to give a negative effect on load-bearing capacity is not expected



#### Comparison of design values with properties after 24 years of use

- For dimensioning of structures made of GFRP creep and aging effects shall be taken into account at least via the following influencing factors:
  - A<sub>1</sub>: Duration of exposure
  - A<sub>2</sub>: Media influence
  - A<sub>3</sub>: Ambient temperature
- For determination of the total reduction factor for strength and stiffness, the following approach was used, with the DIN 18820-2 valid during the construction of the cooling tower:

$$A_F = A_{F1} \times A_2 \times A_3$$
 or  $A_E = \times A_2 \times A_3$ 

#### • For construction profiles of company Fiberline A/S the following factors can be set for service life of 24 years:

 $A^{F} = A^{F}_{1} \times A_{2} \times A_{3} = 1,3 \times 1,2 \times 1,1 = 1,72$ 

 $A^{\rm E} = \sqrt{A_1^{\rm E}} \times A_2 \times A_3 = 1,13 \ge 1,2 \ge 1,13 = 1,50$ 

 $A^{\rm F} = A^{\rm F}_1 \times A_2 \times A_3$  = 1,9 x 1,2 x 1,1 = 2,50

 $A^{\text{E}} = \sqrt{A_1^{E}} \times A_2 \times A_3 = 1,38 \times 1,2 \times 1,1 = 1,82$ 

Strength parallel to profile direction:

Stiffness parallel to profile direction:

Strength perpendicular to profile direction:

Stiffness perpendicular to profile direction :



COMPOS



- Application clean gas channel (CGC) in coal power plant
- Use of GFRP since 1980
- GFRP have been used successfully applications with increased corrosive stress such as media pipes
- Issue: bubbles inside CGC
  - Root cause: formed in intermediate layers, when the substrate laminate is wrapped on a previously fully cured chemical protective layer
  - Mechanism: different coefficients of thermal expansion + thermal expansion + expansion of penetrated liquid
  - Additional effect: If insufficient adhesion between the layers, cavities are formed (bubbles on the surface)





Bubbles inside clean gas channel (CGC)



Cross section of bubble



Clean gas channel (CGC) in brown coal power plant



**A**rplus<sup>●</sup>

![](_page_14_Picture_2.jpeg)

![](_page_15_Picture_1.jpeg)

![](_page_15_Figure_2.jpeg)

![](_page_15_Picture_3.jpeg)

#### Comparison of design values with properties after 20 years of service life

- Following characteristics were set for the laminate with given lay-up structure:
  - Bending modulus (circumferential direction):
  - Bending modulus (longitudinal direction):
  - Bending strength (circumferential direction):
  - Bending strength (longitudinal direction):
  - Creep tendency (longitudinal direction):
- Factors of influence: A1 (long-term load) = 1,4

A2 (media) = 1,33

![](_page_16_Picture_10.jpeg)

 $\rightarrow$  total reduction factor of A<sub>1</sub> x A<sub>2</sub> x A<sub>3</sub>= 2.51 can be derived A3 (temperature) = 1,35

20.000 MPa

11.000 MPa

420 MPa

145 MPa

< 8%

#### $\rightarrow$ Calculation of the reduction using factors of influence, CGC after 175,000 operating hours: circumferential direction

Property	After 175.000 h	Reduction A <sub>1</sub> *A <sub>2</sub> according to 175.000 h	Reduction A <sub>3</sub>	Reduction $A_1 * A_2 * A_3$ according to 175.000 h	Comparison
E <sub>F</sub> [MPa]	28.099	0,71	1,35	0,96	<mark>&lt; 2,51</mark>
σ <sub>F</sub> [MPa]	276,4	1,52	1,35	2,05	<mark>&lt; 2,51</mark>

#### longitudinal direction

Property	After 175.000 h	Reduction A <sub>1</sub> *A <sub>2</sub> according to 175.000 h	Reduction A <sub>3</sub>	Reduction $A_1^*A_2^*A_3$ according to 175.000 h	Comparison
E <sub>F</sub> [MPa]	17.887	0,61	1,35	0,82	<mark>&lt; 2,51</mark>
σ <sub>F</sub> [MPa]	134,6	1,08	1,35	1,46	<mark>&lt; 2,51</mark>

![](_page_16_Picture_16.jpeg)

#### Approach for Failure Determination during Fatigue Test

#### **General material behaviour of FRP**

![](_page_17_Figure_2.jpeg)

![](_page_17_Picture_3.jpeg)

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![](_page_17_Picture_5.jpeg)

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DRESDEN

1...UD in 0° 2...Multiaxial (0°/90°) 3...BIAX

### Approach for Failure Determination during Aging Test

![](_page_18_Picture_1.jpeg)

#### Ultrasonic testing to estimate the fatigue behavior

![](_page_18_Picture_3.jpeg)

![](_page_18_Picture_4.jpeg)

Schematic layout of the test points and test setup

![](_page_18_Picture_6.jpeg)

Coupon sample with NDT after 500 load cycles

![](_page_18_Picture_8.jpeg)

Prüfpunkt	1	2	3	4	5	x	
Bauteildicke in mm	3,20	3,15	3,20	3,10	3,15	= 3,15	
Anzahl Zyklen n	Schallgeschwindigkeit in m/s						
0	3054	3005	2874	2924	2941	2960	
50	3036	2982	2874	2924	2924	2948	
100	3030	3005	2858	2907	2930	2946	
150	3030	2964	2858	2913	2918	2937	
250	3000	2964	2842	2907	2918	2926	
500	3005	2958	2842	2891	2891	2917	
1000	2858	2880	2794	2863	2800	2839	
1500	2634	2647	2652	2625	2567	2625	
1805		2549	2615	2515	2486	2541	
2000							
2500							
3700							
5200							

![](_page_18_Figure_10.jpeg)

![](_page_18_Picture_11.jpeg)

### Approach for Failure Determination during Aging Test

![](_page_19_Picture_1.jpeg)

Change of ultrasonic testing signal due to increasing number of inter-fibre failures and growing delamination in the specimen

#### 0 load cycles

500 load cycles

43.08

![](_page_19_Picture_4.jpeg)

Sa3.14

a2.11 Ra-10.8 Da2.31 ha6.0

BAS SEND EMPF ABLE BBLE

**BILD-BR** 

× 3005

ABB-ANF

PK-VORL

11.45

0.00

5.00

![](_page_19_Picture_5.jpeg)

#### 1.000 load cycles

44.08 : Sa3.13	BILD-BR	
1-0 : 1	5.00	
	C	
	× 2858%	
	ABB-ANF	
	0.00	
A	PK-VORL	
	11.45	
Pa2.11 Ra-10.8 Ba2.31 h	a6.4 000	
BAS ISEND EMPF a	BLE BBLE	

#### 1.500 load cycles

![](_page_19_Figure_9.jpeg)

![](_page_19_Picture_10.jpeg)

![](_page_19_Picture_11.jpeg)

### Approach for Failure Determination during Aging Test

![](_page_20_Picture_1.jpeg)

- Evaluation of the material condition and derivation of predictions on the remaining service life
- Increasing operational safety through recurring monitoring or permanent monitoring in critical areas

#### Suitable NDT methods for structural monitoring of GFRP components:

- Ultrasonics (UT)
- Guided waves (AU)
- Acoustic emission (AE)

#### Requirement:

- Numerous test data for different material
- Consideration of load types / load direction
- Data basis for the evaluation of NDT-images

![](_page_20_Picture_12.jpeg)

**Acplus**<sup>th</sup>

![](_page_20_Picture_13.jpeg)

![](_page_21_Picture_0.jpeg)

![](_page_21_Picture_1.jpeg)

- 1. With an optimised choice of reinforcement fibre, layer structure, matrix and, if necessary, fillers to meet the requirements of the application, construction projects with the highest demands can be realised over a long period of time (>25 years).
- 2. Applications of GRP from all industries and technologies allow us to expect service lives of 50 100 years.
- 3. The actual degradation of the material depends not only on the loads and the choice of material, but also on the manufacturing quality (degree of cross-linking) and on the assembly (residual stress).
- 4. I.e. the degradation factors required for the design are not purely theoretically determinable factors! They can differ greatly in practice, because they depend on the manufacturing quality.
- 5. The use of "witness samples" help to determine the long-term behaviour more reliably.
- 6. Non-destructive testing methods are suitable for structural monitoring and evaluation of the material condition.

## With our materials and pultruded profiles, we are well positioned in the competition of materials.

![](_page_21_Picture_9.jpeg)

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![](_page_22_Picture_3.jpeg)

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